

## SUPPLEMENTAL TYPE CERTIFICATE

### 10076572

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1  
94348 ATTING  
GERMANY

EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** US A7SO

**Type Certificate Holder:** PIPER AIRCRAFT, INC.

**Type:** PA-34

**Model:** PA-34-200 (SENECA)

**Description of Design Change:**

Installation of MTV-12-B-C-F/CF(L)183-59( ) Propeller on PA-34-200(Seneca)

**EASA Certification Basis:**

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

- The following paragraph(s) at a later amendment: CS 36 Amdt 5; FAR 36-31.

This certificate/approval involves a change with an impact on the noise database and/or emission levels. The environmental protection requirements are amended as follows: CS 36 Amdt 5, FAR 36-31.

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

**Cologne, Germany, 28 May 2021**



**Manfred REICHEL**  
**Section Manager**

**Small Aircraft, Balloons & Airships Section**





**Associated Technical Documentation:**

- According to MT-Propeller Master Documentation List No. E-3549 Rev. 0
- To be operated in accordance with MT-Propeller AFM-S Document No. E-3551 Rev. 0
- To be installed i.a.w. Installation Instruction Doc. No. E-3552 Rev. 0
- To be maintained i.a.w. MT-Propeller ICA Document No. E-3553 Rev. 0

or later revisions of the above listed document(s) approved/accepted under the EASA system.

**Limitations/Conditions:**

According limitation section of AFM-S.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -